

SUPPLEMENTAL TYPE CERTIFICATE

10086773

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ALPINE AEROTECH LP

1260 INDUSTRIAL ROAD
WEST KELOWNA BC V1Z 1G5
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.R.106

Type Certificate Holder: BELL TEXTRON, INC.

Type: Bell 212/412

Model: 212

412

412EP

Original STC Number: SH11-23

Description of Design Change:

Relocation of Engine Oil Pressure Transducer.

EASA Certification Basis:

The Type and OSD Certification Bases for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

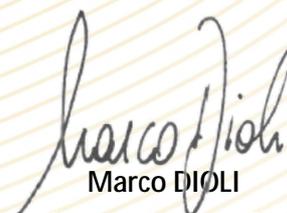
Associated Technical Documentation:

- Master Data List AAL-214-034-001 Rev. NI dated 2011-04-01.
- Installation Instruction in accordance with ICA document AAL-214-035-701 Rev. NI dated 2011-05-04.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 27 March 2025



Marco DIOLI
Section Manager Heavy Rotorcraft

-Instructions for Continued Airworthiness AAL-214-035-701 Rev. NI dated 2011-05-04.
Or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

